

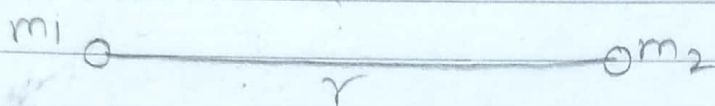
SBG STUDY

01/06/17

Chapter:

Gravitation

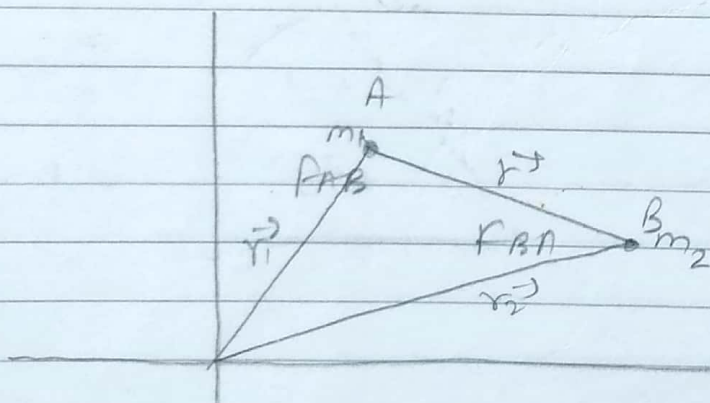
* Newton law of Gravitation:



$$F = \frac{G m_1 m_2}{r^2}$$

(G = universal Grav. Constant)

- * Gravitational force act along line of joining
- * It is a Conservation force
- * It is always attracted
- * It is Independent of Median



$$\vec{r}_{12} = \vec{r}_2 - \vec{r}_1$$
$$|\vec{r}_{12}| = r$$

$$\frac{G m_1 m_2}{r^2}$$

$$\vec{F}_{AB} = -\vec{F}_{BA}$$

$$\vec{F}_{BA} = \frac{G m_1 m_2}{r^2} (-\hat{r})$$

$$\vec{F}_{BA} = -\frac{G m_1 m_2}{r^2} \hat{r}$$

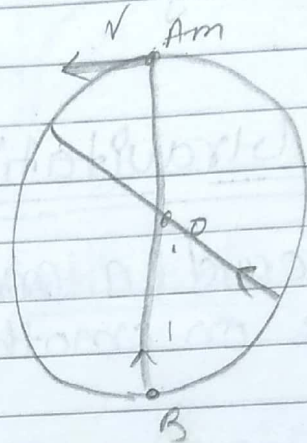
$$\vec{F}_{AB} = -\vec{F}_{BA}$$

Que: Two particles of mass (m) are moving on a circle of radius (R) due to mutual gravitation force. find speed of each particle.

Ans:

$$f = \frac{Gm^2}{4R^2} = \frac{mv^2}{R}$$

$$\frac{Gm}{4R} = v^2$$



Que: Three particles are moving on a circle of radius (r) such that they always formed an equilateral triangle. find speed of each mass.

Ans:

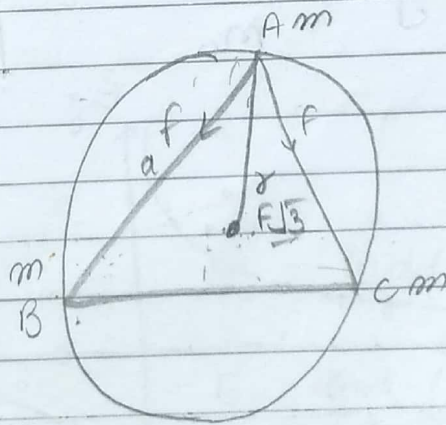
$$\frac{a}{\sqrt{3}} = r$$

$$a = \sqrt{3}r$$

$$f = \frac{Gm^2}{a^2}$$

$$f\sqrt{3} = \frac{mv^2}{r}$$

$$\frac{Gm^2\sqrt{3}}{a^2} = \frac{mv^2}{r}$$



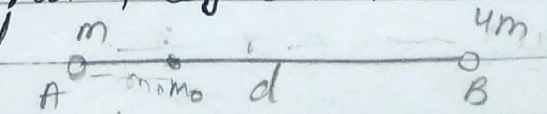
$$\frac{a}{\sqrt{3}} = r$$

$$a = \sqrt{3}r$$

Que: mass m and $4m$ are separated by distance (d). where a third mass (m_0) should be placed show that net force on it becomes zero.

$$\frac{Gmm_0}{r^2} = \frac{4Gmm_0}{(d-r)^2}$$

$$r = \frac{d}{3} \text{ Ans.}$$

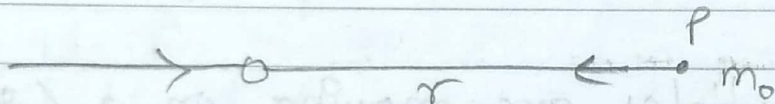


* Equilibrium is unstable along line of joining.

* Stable perpendicular along line of joining.

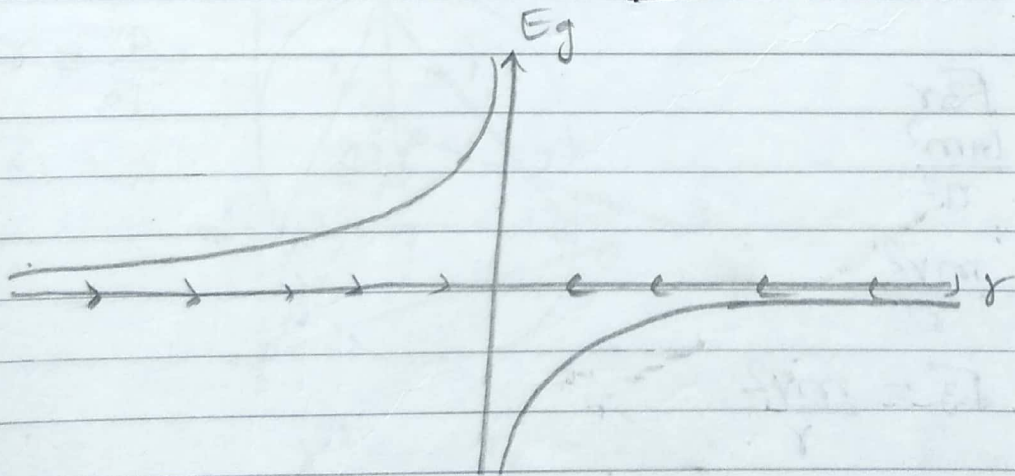
* Gravitational field $\therefore (E_g)$

It is a region around a mass in which it applies gravitational force on another mass.

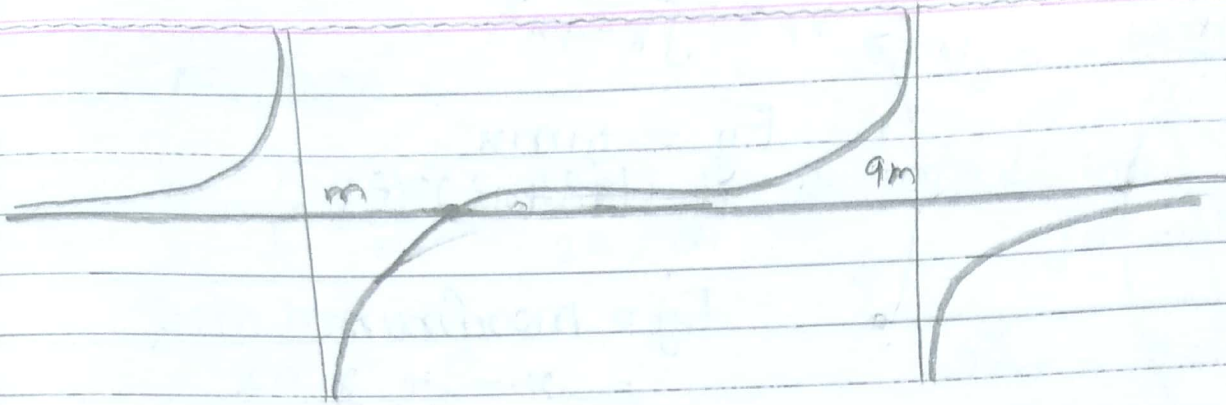


$$F = \frac{Gmm_0}{r^2}$$

$$E_g = \frac{F}{m_0} = \boxed{E_g = \frac{Gm}{r^2}} \text{ (radially inward)}$$

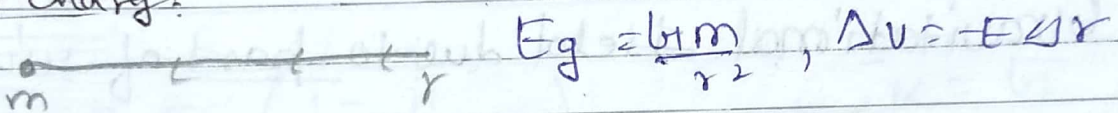


Ques:

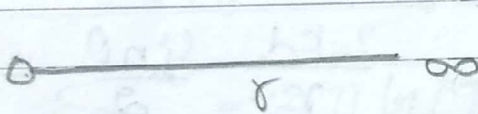


* Gravitational field and potential due to different bodies:

(i) Point charge:

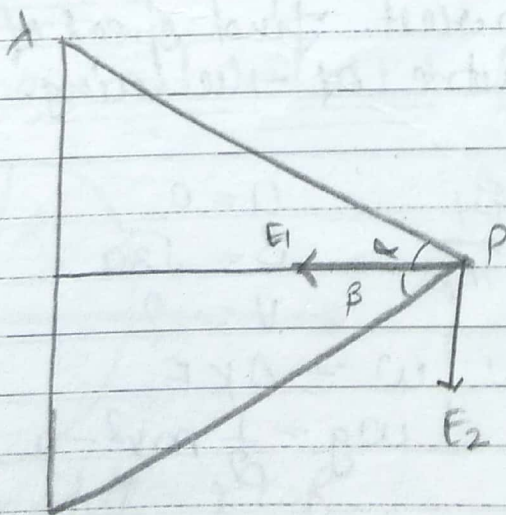


$$E_g = \frac{GM}{r^2}, \Delta V = -E dr$$



$$V_g = -\frac{GM}{r}, V = 0$$

(ii) Uniform Rod:



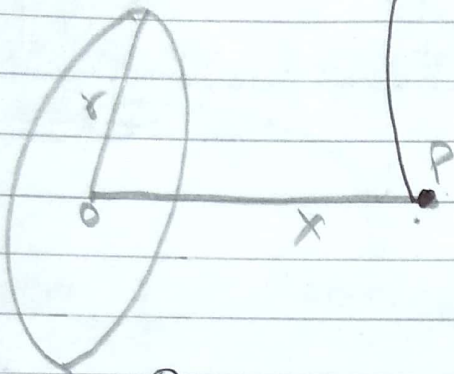
$$E_2 = \frac{G \rho d}{r} (\cos \beta - \cos \alpha)$$

$$E_1 = \frac{G \rho d}{r} (\sin \alpha + \sin \beta)$$

long rod $\Rightarrow E_1 = \frac{2G \rho d}{r}, \alpha = \beta = 0$

$$E_2 = 0$$

(iii) Ring!



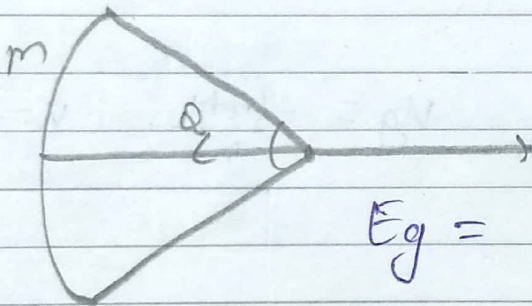
$$V_P = -\frac{Gm}{\sqrt{r^2 + x^2}}$$

$$E_g = \frac{Gm x}{(r^2 + x^2)^{3/2}}$$

$$E_g = \text{maximum} \\ = x = \pm \frac{r}{\sqrt{2}}$$

Potential: $V_0 = -\frac{Gm}{r}$

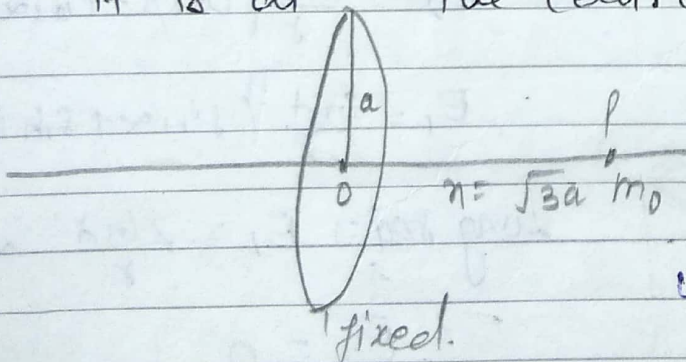
* Gravitational field due to part of ring!



$$\frac{2kd \sin \theta}{r}$$

$$E_g = \frac{2Gd}{r} \cos \theta$$

Que! m_0 is released from rest. Find speed of m_0 when it is at m the centre of the ring.



$$u = 0$$

$$s = \sqrt{3}a$$

$$v = ?$$

$$W = \Delta KE$$

$$W_g = \frac{1}{2} mv^2 - 0$$

$$= -m_0(V_0 - V_P) = \frac{1}{2} mv^2$$

$$W_{\text{elec}} = -q \Delta V$$

$$W_{\text{grav}} = -m \Delta V$$

↓

$$-m_0 \left(\frac{GM}{a} \right)$$

$$V_0 = -\frac{GM}{a}$$

$$-m_0 \left(-\frac{GM}{a} + \frac{GM}{2a} \right) = \frac{1}{2} m_0 v^2$$

$$V_p = -\frac{GM}{\sqrt{a^2 + 3a^2}} = -\frac{GM}{2a}$$

$$\frac{GM m_0}{2a} = \frac{1}{2} m_0 v^2$$

$$W_{ext} = q \Delta V$$

$$v = \sqrt{\frac{GM}{a}}$$

$$W_{ext} = m \Delta V$$

* DISC:

$$E = \frac{\sigma}{2\epsilon_0} \left(\frac{1-x}{\sqrt{r^2+x^2}} \right)$$

$$K = G$$

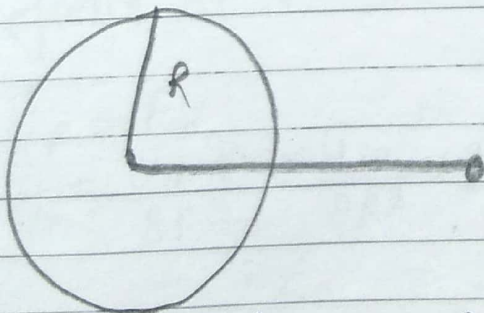
$$\frac{1}{4\pi\epsilon_0} = G$$

$$E_g = -2\pi G \left(\frac{1-x}{\sqrt{r^2+x^2}} \right)$$

$$\frac{1}{\epsilon_0} = 4\pi G$$

* SPHERE

(i) Spherical shell: (Hollow shell).



(i) $r > R$ Outside

$$E_g = \frac{GM}{r^2}$$

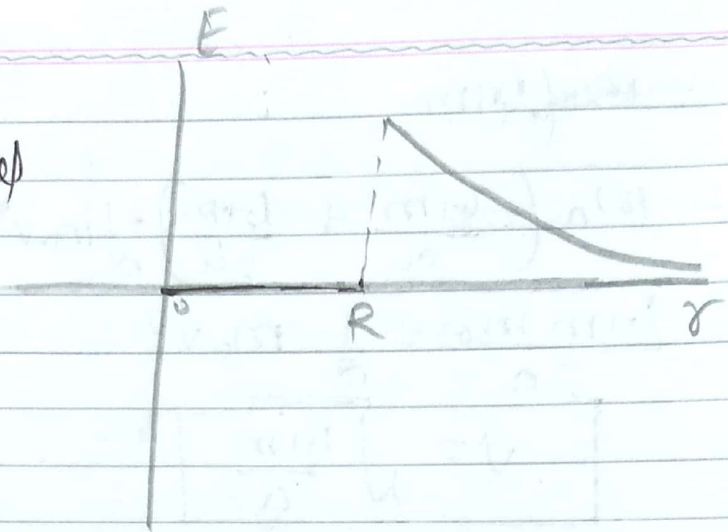
$$V = -\frac{GM}{r}$$

(ii) Inside $r < R$

$$E_g = 0$$

$$V = -\frac{GM}{R}$$

Graph of Spheri'cal shell
(inside)

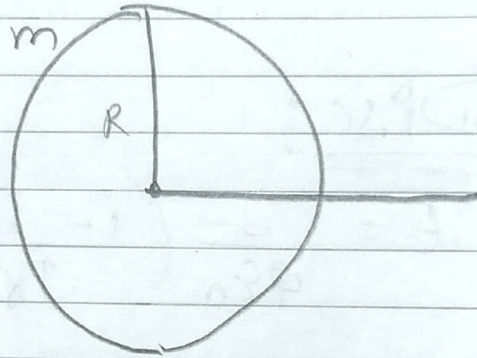


* Solid sphere :

(i) Outside point

$$r > R$$

$$E_g = \frac{Gm}{r^2}$$



$$V_g = \frac{-Gm}{r}$$

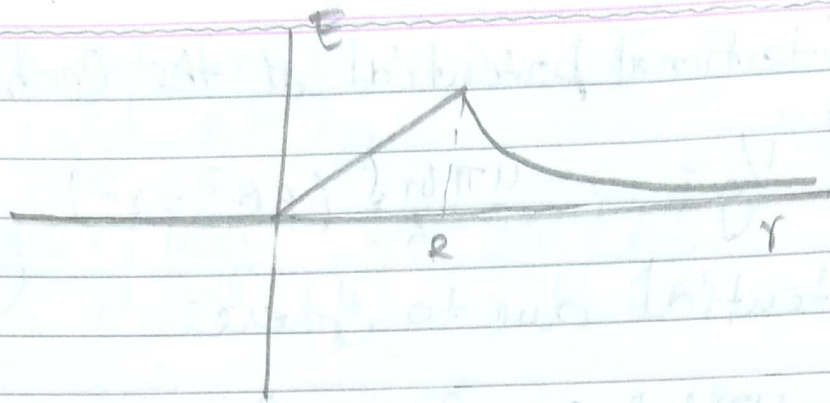
(ii) Inside point : ($r < R$)

$$E_g = \frac{4\pi G}{3} \rho r$$

$$\int \frac{dr}{r^2}$$

$$= \frac{4\pi G}{3} \frac{m}{\frac{4}{3}\pi R^3} \cdot r$$

$$E_g = -\frac{Gm r}{R^3} \quad \text{in term of density.}$$



$$\frac{\int}{6\epsilon_0} (3R^2 - r^2)$$

$$V_g = \frac{4\pi G}{6} (3R^2 - r^2) \quad \text{in term of density}$$

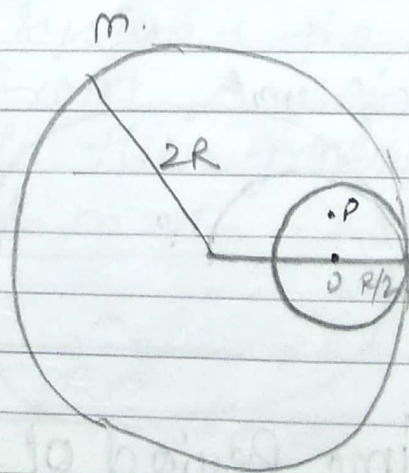
$$V_g = \frac{4\pi G}{6} \times \frac{m}{\frac{4\pi}{3} R^3} (3R^2 - r^2)$$

$$V_g = \frac{Gm}{2R^3} (3R^2 - r^2) \quad \text{in term of mass.}$$

Que: Find Gravitational field at P

Ans: $E_p = E_0$
 $E_1 = \frac{Gm}{8R^3} = \frac{Gm}{8R^3} \times \frac{3R}{2}$

$$E_1 = \frac{3Gm}{16R^2}$$



(ii) Find Gravitational potential at the Centre of Cavity

Ans! $V_g = -\frac{4\pi G \rho}{6} (3R^2 - r^2)$ $\left(\rho = \frac{m}{\frac{4\pi}{3}R^3}\right)$

Potential due to sphere

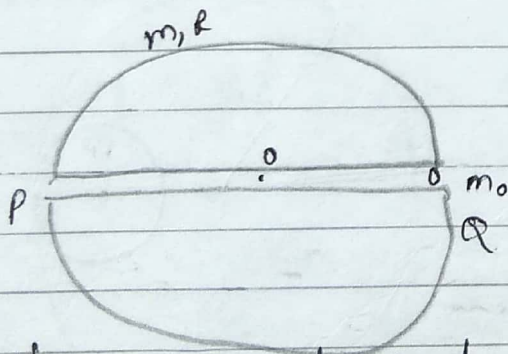
$$V_1 = -\frac{4\pi G \rho}{6} \left(12R^2 - \frac{9R^2}{4}\right)$$

$$V_2 = -\frac{4\pi G \rho}{6} \left(\frac{3R^2}{4} - 0\right)$$

$$V_{01} = V_1 - V_2$$

~~Ques~~

Ques!



Find speed of mass at the centre of sphere

Ans! $-m_0(v_0 - v_a) = \frac{1}{2}m_0v^2$

$$E_p = \frac{Gm m_0}{R^3} r$$

$$F = m_0 E_p$$

$$F = \left(\frac{Gm m_0}{R^3}\right) r$$

Find time Period of SHM:

Ans $V_{max} = A\omega$

$$V_{max} = R\omega$$

$$T = \frac{2\pi}{\omega}$$

O - Centre
& surface

Conduct
H.V

$R = 6 \times 10^6$

Earth Time period

$= 84.6 \text{ m.}$

$$a = \left(\frac{GM}{R^3} \right) r$$

$$\omega = \sqrt{\frac{GM}{R^3}} \quad T = 2\pi \sqrt{\frac{R^3}{GM}}$$

Time Period of Earth = 84.6 m.

(J.M) Que.

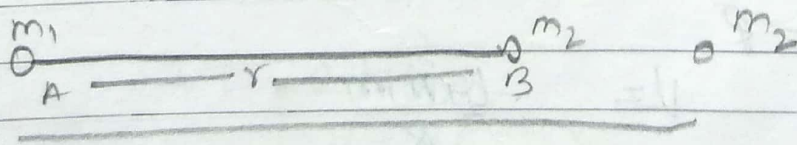
(15) $m = \frac{4\pi r^3}{3} \rho$ $gE = \int \frac{4\pi r^3 \rho}{3} g$

$$6\pi r^2 v = \int \frac{4\pi r^3 \rho}{3} g$$

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(2)

~~X~~ Gravitational potential Energy

Gravitational
p.e of two point mass system is the work done to take mass from a reference point to infinity in the gravitational field of another point mass.



$$F = \frac{GM_1 M_2}{r^2}$$

$$W = - \int_{r_0}^{\infty} \frac{GM_1 M_2}{r^2} dr = - GM_1 M_2 \left(\frac{1}{\infty} - \frac{1}{r_0} \right)$$

$$W = \frac{Gm_1 m_2}{r_0}$$

$$W_{\text{ext}} = -\Delta U = -q \Delta V$$

$$W \Delta U = \frac{Gm_1 m_2}{r_0}$$

$$U_f - U_i = \frac{Gm_1 m_2}{r_0}$$

$$0 - U_0 = \frac{Gm_1 m_2}{r_0}$$

$$U_0 = \frac{Gm_1 m_2}{r_0}$$

$$PE = mV$$

$$V = -\frac{Gm_1}{r}$$

$$PE = m_2 V$$

$$= -\frac{Gm_1 m_2}{r}$$

$$W_{\text{grav.}} = -\Delta U = -m \Delta V$$

$$W_{\text{ext}} = m \Delta V = \Delta U$$

Que: Find potential energy of system if

(a)

$$r > R$$

$$V = \frac{Gm}{r}$$

$$U = mV = -\frac{Gm_1 m}{r}$$

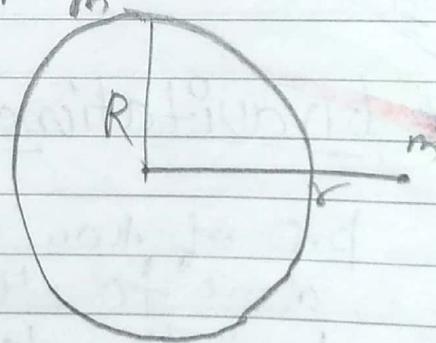
b) $r = R$

$$V = -\frac{Gm_1 m}{R}$$

c) $r < R$

$$V = \frac{GM}{2R^3} (3R^2 - r^2)$$

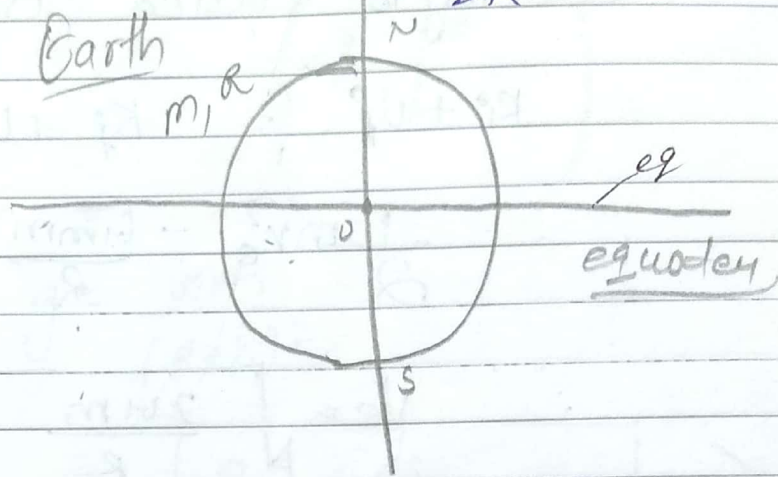
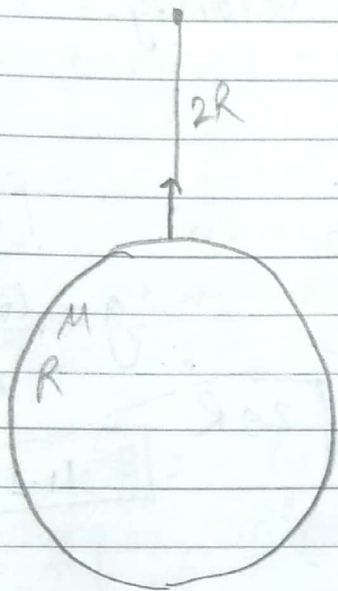
$$U = \frac{Gm_1 m}{2R^3} (3R^2 - r^2)$$



* Potential at centre :-

$$V_{\text{centre}} = -\frac{3GMm}{2R}$$

Que :-



An object of mass m projected vertically upward from earth surface so, that it reaches to height $2R$. Find initial speed of the particles.

Ans :-

$$PE_i + KE_i = -\frac{GMm}{3R} = \frac{1}{2}mv^2$$

$$PE_f + KE_f = KE_i + PE_i$$

$$\frac{1}{2}mv^2 - \frac{GMm}{R} = 0 - \frac{GMm}{3R}$$

$$v = 2\sqrt{\frac{GM}{3R}}$$

* Escape Velocity :- When an object is projected from earth surface so that it does not return to the earth surface.

OR
 It leave gravitational field of earth that velocity is called escape velocity.

$$K_i^0 + U_i^0 = K_f + U_f$$

$$\frac{1}{2} m v_e^2 - \frac{G m m}{R} = 0$$

$$v_e = \sqrt{\frac{2 G m}{R}} = \sqrt{2 g R}$$

$$g = \frac{G m}{R^2}$$

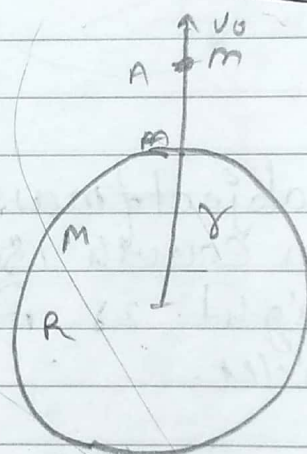
$$= 11.2 \text{ km/sec.}$$

Ques: find v_0 so that it's escape from gravitational field

Ans: $K_i^0 + U_i^0 = K_f + U_f$

$$\frac{1}{2} m v_0^2 - \frac{G m m}{r} = 0 \text{ to}$$

$$v = \sqrt{\frac{2 G M}{r}} = \sqrt{2 g r}$$



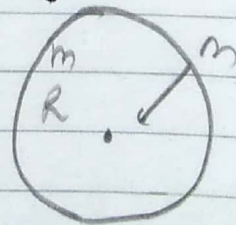
* Acceleration due to gravity: (g).

$$f = \frac{G m m}{R^2}$$

$$a = \frac{f}{m}$$

$$g = \frac{f}{m}$$

$$g = \frac{G m}{R^2}$$

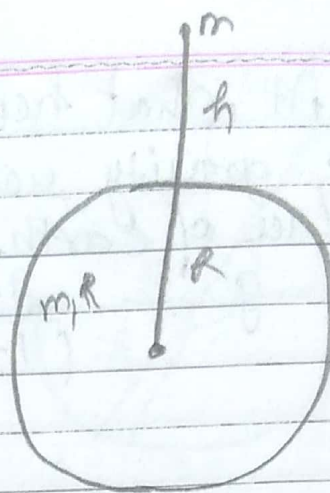


(ii) Above Earth Surface:

$$r = R + h$$
$$F = \frac{GMm}{r^2}$$

$$g = \frac{F}{m}$$

$$g = \frac{Gm}{r^2} \Rightarrow g = \frac{Gm}{(R+h)^2}$$



$$g = \frac{Gm}{R^2 \left(1 + \frac{h}{R}\right)^2}$$

$$g = \frac{g_s}{\left(1 + \frac{h}{R}\right)^2}$$

Pf $h \ll R$

$$g = g_s \left(1 + \frac{h}{R}\right)^{-2}$$

$$g = g_s \left(1 - \frac{2h}{R}\right)$$

$$\frac{g}{g_s} = 1 - \frac{2h}{R} \Rightarrow \frac{2h}{R} = 1 - \frac{g}{g_s}$$

$$\frac{2h}{R} = \frac{g_s - g}{g_s} \Rightarrow \frac{\Delta g}{g_s} = \frac{2h}{R}$$

fraction change.

Ques: 12,

S = 1, 2, 6, 7, 9, 13, 15,

At what height above earth surface acc. due to gravity remains 25% of its value at the surface of earth.

Ans: $g = \frac{g_s}{\left(\frac{1+h}{R}\right)^2} \Rightarrow \frac{g_s}{4} = \frac{g_s}{\left(\frac{1+h}{R}\right)^2}$

$$\left(\frac{1+h}{R}\right)^2 = 4 \Rightarrow 1 + \frac{h}{R} = 2$$

$$h = R$$

Ques: At what height above surface acc. due to gravity changes by one Percent.

Ans: $\frac{\Delta g}{g_s} \times 100 = \frac{2h}{R} \times 100$

$$1 = \frac{2h \times 100}{R}$$

$$h = \frac{R}{200}$$

$$h = 32 \text{ km.}$$

(iii) below earth surface:

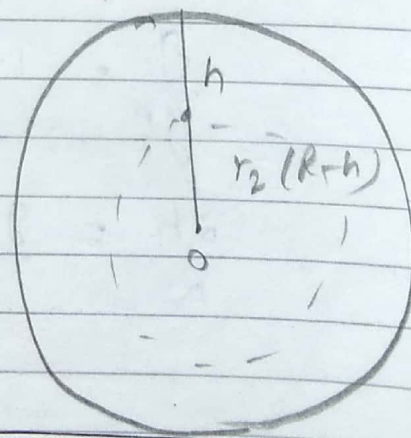
$$E_g = \frac{GM \cdot r}{R^3}$$

$$F = mE_g = \frac{GMm \cdot r}{R^3}$$

$$g = \frac{F}{m} = \frac{GM \cdot r}{R^3}$$

$$g = \frac{GM}{R^3} (R-h)$$

$$g = g_s \left(1 - \frac{R-h}{R}\right)$$



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Ex: 8-2

$$(1) \quad dE = \frac{2k\lambda}{R}$$

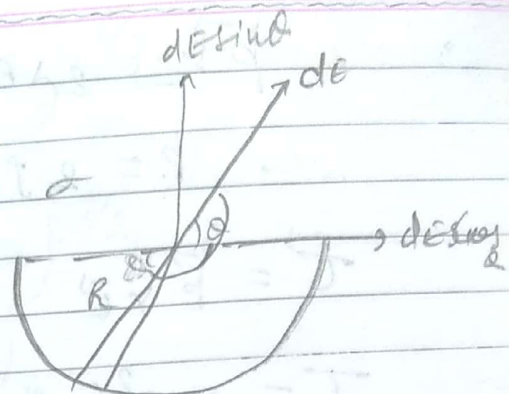
$$= \frac{2k\sigma R d\theta}{R} = 2k\sigma d\theta$$

$$E_y = \int dE \sin\theta$$

$$= 2k\sigma \int_0^\pi \sin\theta d\theta$$

$$= 2k\sigma (\cos\theta) \Big|_0^\pi$$

$$= 4k\sigma R$$



$$a) \quad dq = \lambda R d\theta$$

$$dp = dq \cdot 2R$$

$$= \lambda R \cdot 2R d\theta$$

$$= 2\lambda R^2 d\theta$$

$$p_x = \int dp \cos\theta$$

$$p_x = \int dp \cos\theta$$

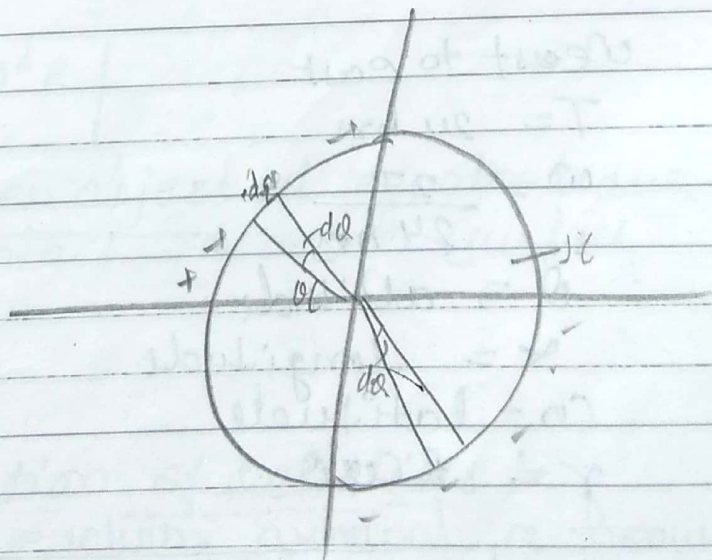
$$= 2\lambda R^2 \int_0^{\pi/2} \cos\theta d\theta$$

$$p_x = 2\lambda R^2$$

$$p_y = \int dp \sin\theta$$

$$= 2\lambda R^2 \int_0^{\pi/2} \sin\theta d\theta$$

$$= 2\lambda R^2$$



acc. = 0 inertia

Earth - Non-inertial frame

$$\vec{p} \Rightarrow 2\lambda R^2 (-\hat{i} + \hat{j})$$

$$p = 2\lambda R^2$$

$$\vec{\tau} = \vec{p} \times \vec{E}$$

$$\tau = 2\lambda R^2 (-\hat{i} + \hat{j}) \times E_0 \hat{i}$$

$$\tau = -2\lambda R^2 E_0 \hat{j}$$

$$a = R\alpha$$

$$f = m a$$

$$\tau = I \alpha$$

$$\frac{f}{\tau} = \frac{m R}{I}$$

15]

$$r = \sqrt{R^2 + h^2}$$

$$f = \frac{\sigma^2}{2\epsilon_0} \times \pi (R^2 - h^2)$$

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(iv) Variation in acc. due to gravity due to rotation of earth:

West to east

$$T = 24 \text{ hrs}$$

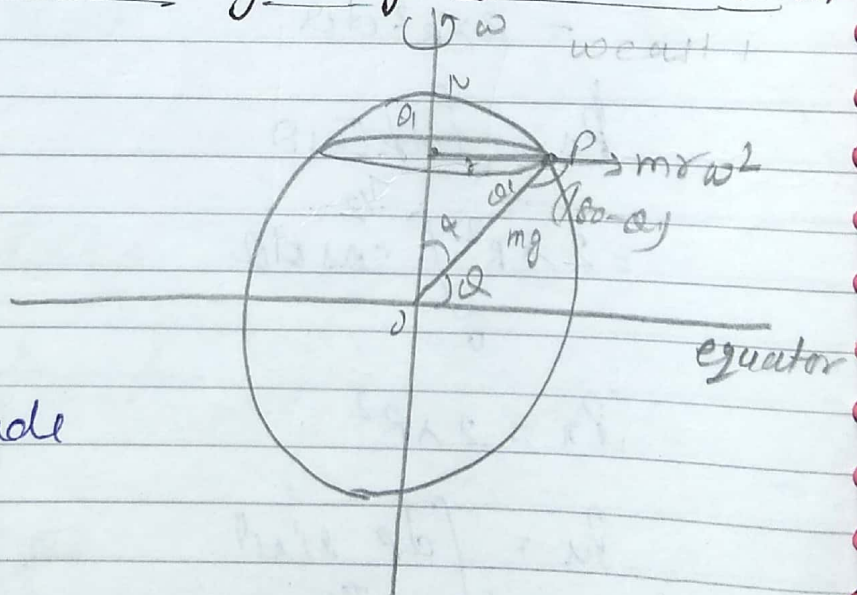
$$\omega = \frac{2\pi}{24 \text{ hrs}}$$

θ = altitude

λ = longitude

Co-latitude

$$r = R \cos \theta$$



$$F_{\text{Net}} = \sqrt{(mg)^2 + (\underbrace{m r \omega^2}_{\text{neglected}})^2 + 2mg m r \omega^2 \cos(180 - \theta)}$$

$$F_{\text{net}} = (m^2 g^2 - 2m^2 g r \omega^2 \cos \theta)^{1/2}$$

$$f_{\text{net}} = mg \left(1 - \frac{2R\omega^2 \cos^2 \theta}{g} \right)^{1/2}$$

$$f_{\text{net}} = mg \left(1 - \frac{2R\omega^2 \cos^2 \theta}{g} \right)^{1/2}$$

$$f_{\text{net}} = mg \left(1 - \frac{\omega^2 R \cos^2 \theta}{g} \right)$$

$$mg' = mg \left(1 - \frac{\omega^2 R \cos^2 \theta}{g} \right)$$

$$g' = g - \omega^2 R \cos^2 \theta$$

Poles - North Pole

$$\theta = 90^\circ$$

$$g' = g = 9.8 \text{ m/s}^2$$

equator $\theta = 0$

$$g' = g - \omega^2 R$$

If $g' = 0$, then object at equator leave
Contact to earth surface

$$\omega = \sqrt{\frac{g}{R}}$$

* Satellites & motion of satellites?

A light object revolving around a heavy object due to gravitational force only called satellite.

* Artificial Satellites:

radius of orbit (r)

$$r = R + h$$

Or

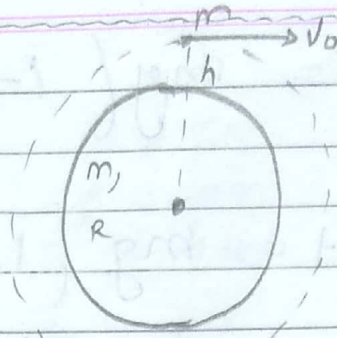
↳ orbital velocity:

$$\left[g = \frac{GM}{r^2} \right]$$

$$v_0 = \sqrt{\frac{GM}{r}}$$

$$\frac{GMm}{r^2} = \frac{mv_0^2}{r}$$

$$v_0 = \sqrt{gr}$$



* Time Period of Satellites:

$$T = \frac{2\pi r}{v}$$

$$v = \sqrt{gr}$$

$$T_1 = \frac{2\pi r}{v_0} =$$

$$= \frac{2\pi r^{3/2}}{\sqrt{GM}}$$

$$= T \propto r^{3/2}$$

$$T^2 \propto r^3$$

Time Period of 2 satellites

$$\boxed{\frac{T_1}{T_2} = \left(\frac{r_1}{r_2} \right)^{3/2}}$$

* Energy of Satellites:

$$T.E = K.E + P.E$$

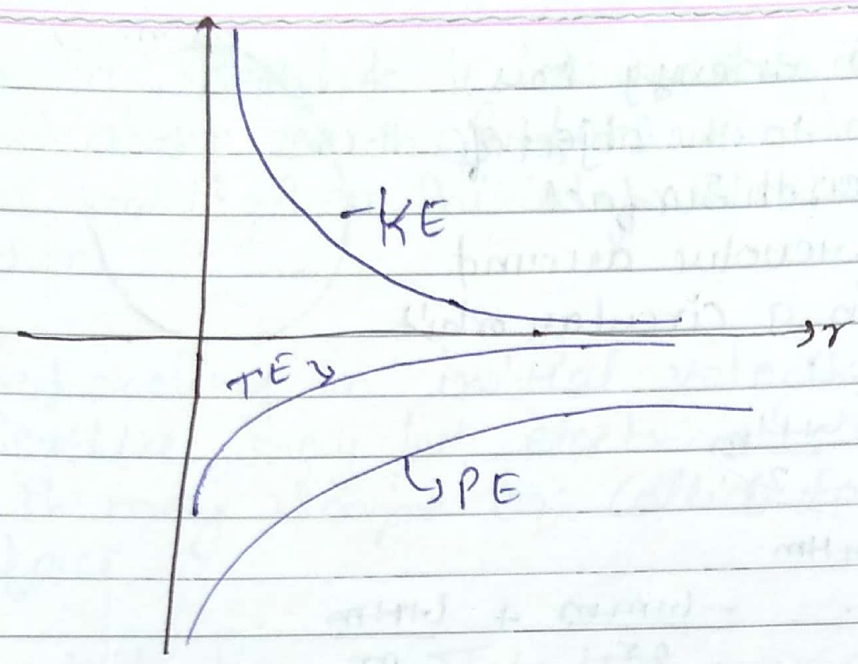
$$K.E = \frac{1}{2} m v_0^2$$

$$K.E = \frac{GMm}{2r}, \quad P.E = -\frac{GMm}{r}$$

$$T.E = \frac{GMm}{2r} - \frac{GMm}{r}$$

$$\boxed{T.E = -\frac{GMm}{2r} = \frac{P.E}{2}}$$

bra f, m x F, r



radius $r = PE \downarrow$
or all r

$$g_{app} = g - \omega^2 r = 0$$

* Angular momentum of satellite (L) =

$$L = mvr$$

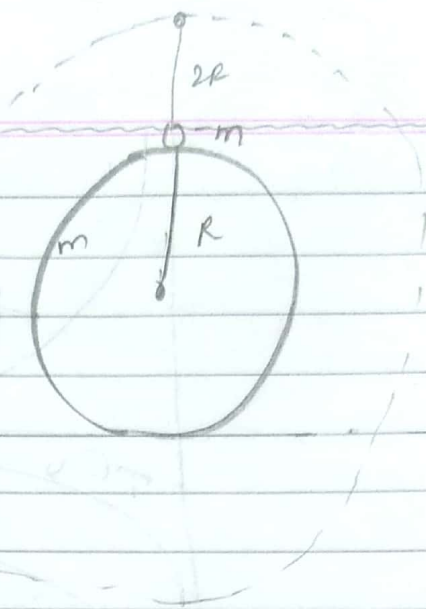
$$L = m \cdot v \cdot r$$

* Work done to change orbit of satellite!
from r_1 to r_2 :

$$\text{Work done} = TE_f - TE_i$$

$$W = \left(-\frac{GMm}{2r_2} + \frac{GMm}{2r_1} \right)$$

Ques] How much energy has to be given to the object of mass (m) on earth surface show that it revolves around the earth in a circular orbit of radius $3R$.



Ans] $E = \frac{-GMm}{2R}$

$= -GMm$

$E_{total} = \frac{-GMm}{2R_2} + \frac{GMm}{2R_1}$

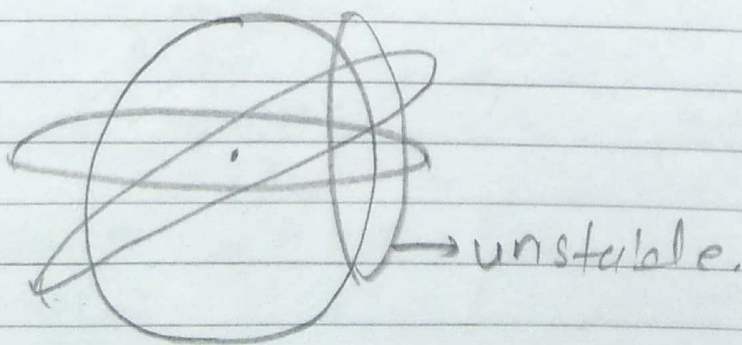
$K_i + U_i = K_f + U_f$

$K_i - \frac{GMm}{R} = -\frac{GMm}{6R}$

$K_i = \frac{GMm}{R} - \frac{GMm}{6R}$

$K_i = \frac{5GMm}{6R} = \frac{5}{6} mg_s R$

* Launching of Satellite: when a satellite revolves around a earth orbit of satellite is called stable orbit. if centre of orbit coincide to the centre of earth.

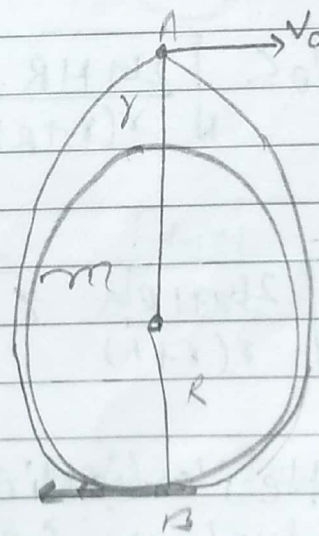


When a satellite is launched in a particular orbit above earth surface it is taken to that point and given initially velocity in horizontal direction.

* Depending on initial velocity orbit of satellite may be elliptical circular or it may escape or collide to the earth surface.

* In this process total energy and angular momentum remains conserved.

Ques: Find minimum value of v_0 so that it does not collide to the earth surface



$$T_{EA} = T_{EB}$$

$$\frac{1}{2} m v_0^2 - \frac{GMm}{r} = \frac{1}{2} m v^2 - \frac{GMm}{R}$$

$$L_A = L_B$$

$$m v_0 r = m v R$$

$$v = \frac{v_0 r}{R}$$

$$\frac{v_0^2}{2} - \frac{GM}{r} = \frac{v^2}{2} - \frac{GM}{R}$$

$$\frac{GM}{R} - \frac{GM}{r} = \frac{v^2 - v_0^2}{2}$$

$$2GM \left(\frac{1}{R} - \frac{1}{r} \right) = v_0^2 \left(\frac{r^2}{R^2} - 1 \right)$$

$$2GM \left(\frac{r-R}{Rr} \right) = v_0^2 \left(\frac{r^2-R^2}{R^2} \right)$$

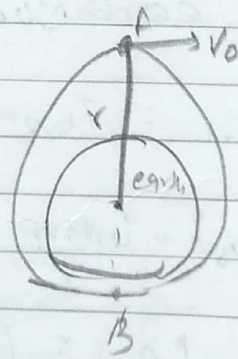
$$\frac{2GM}{r} = v_0^2 \left(\frac{r+R}{R} \right)$$

$$v_0 = \sqrt{\frac{2GM R}{r(r+R)}}$$

i) if $v_0 < \sqrt{\frac{2GM R}{r(r+R)}}$

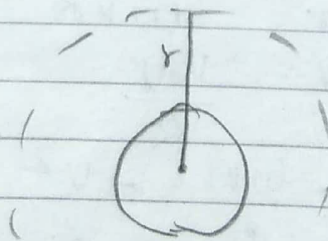
Sattelite collide to earth surface.

ii) if $\sqrt{\frac{2GM R}{r(r+h)}} < v_0 < \sqrt{\frac{GM}{r}}$

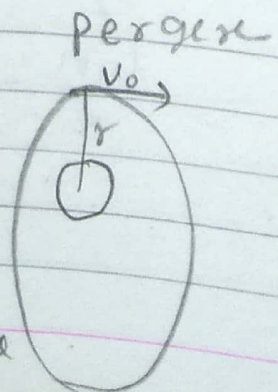


Sattelite collide to earth surface (Aparak).

iii) if $v_0 = \sqrt{\frac{GM}{r}}$



iv) $\sqrt{\frac{GM}{r}} < v_0 < \sqrt{\frac{2GM}{r}}$



Aparogee

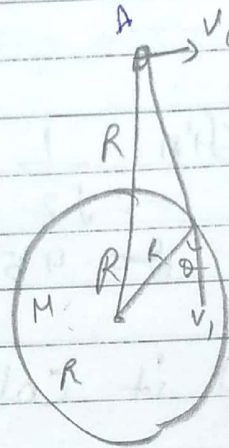
(v) $v_0 = \sqrt{\frac{2GM}{r}}$ - it escape along parabolic path.

(vi) $v_0 > \sqrt{\frac{2GM}{r}}$ It escape along hyperbolic.

Que:

$$v_0 = \sqrt{\frac{GM}{7R}} \text{ will it.}$$

Ans: ~~Will~~ collide to earth surface.



$$v_0 < \sqrt{\frac{2GM}{r}}$$

$$r = 2R$$

$$v = \sqrt{\frac{2GM}{2R \times 3R}} = \sqrt{\frac{GM}{3R}}$$

(ii) Find speed of satellite when it crash into earth.

$$\text{Ans: } K_i + U_i = K_f + U_f$$

$$\frac{1}{2} m v_0^2 - \frac{GMm}{2R} = \frac{1}{2} m v^2 - \frac{GMm}{R}$$

$$v = \sqrt{\frac{8GM}{7R}}$$

(iii) In which direction or At what angle from vehicle it collide to earth surface

Ans! $L_A = L_B$
 $mv_0 2R = mv \sin \theta R$

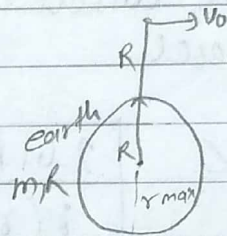
$$\sin \theta = \frac{2v_0}{v} = \frac{2 \sqrt{\frac{GM}{7R}}}{\sqrt{\frac{8GM}{7R}}} = \frac{2}{\sqrt{8}}$$

$$\sin \frac{1}{\sqrt{2}}$$

$$\theta = 45^\circ$$

Ques! will it collide to earth?

↓



$$v = \sqrt{\frac{3GM}{5R}}$$

$$v_0 = \sqrt{\frac{3GM}{5R}}$$

$$= \sqrt{\frac{2GM R}{r(r+R)}} = \sqrt{\frac{2GM R}{2R \times 3R}} = \sqrt{\frac{GM}{3R}}$$

$$v_0 > \sqrt{\frac{GM}{3R}}$$

ii) what will be the path of satellite

$$\sqrt{\frac{GM}{r}} = \sqrt{\frac{GM}{2R}}$$

$$v_0 > \sqrt{\frac{GM}{2R}} \quad v_e = \sqrt{\frac{2GM}{r}} = \sqrt{\frac{4M}{R}}$$

find max. distance of satellite from earth centre.

$$L = m v \cdot r \quad \cancel{K_i} + \cancel{U_i} = \cancel{K_f} + \cancel{U_f}$$

$$= m v$$

$$\cancel{K_i} + U_i = \cancel{K_f} + U_f$$

$$\frac{1}{2} m v_0^2 + =$$

$$m v_0 \cdot 2R = m v r$$

$$v = \frac{2 v_0 R}{r}$$

$$\frac{1}{2} m v_0^2 - \frac{GMm}{2R} = \frac{1}{2} m \frac{4 v_0^2 R^2}{r^2} - \frac{GMm}{r}$$

$$\frac{1}{2} m v_0^2 - \frac{2 m R^2 v_0^2}{r^2} = \frac{GMm}{2R} - \frac{GMm}{r}$$

$$v_0^2 \left(\frac{1}{2} - \frac{2R^2}{r^2} \right) = \frac{GM}{2R} - \frac{GM}{r}$$

$$\frac{3GM}{5R} \left(\frac{1}{2} - \frac{2R^2}{r^2} \right) = \frac{GM}{2R} - \frac{GM}{r}$$

$$\frac{3}{10R} - \frac{6}{5r^2} = \frac{1}{2R} - \frac{1}{r}$$

$$\frac{1}{r} - \frac{6R}{5r^2} = \frac{1}{2R} - \frac{3}{10R}$$

$$\frac{5r - 6R}{5r^2} = \frac{2}{10R} \Rightarrow 5Rr - 6R^2 = r^2$$

$$r^2 - 5Rr + 6R^2 = 0$$

$$(r - 2R)(r - 3R) = 0$$

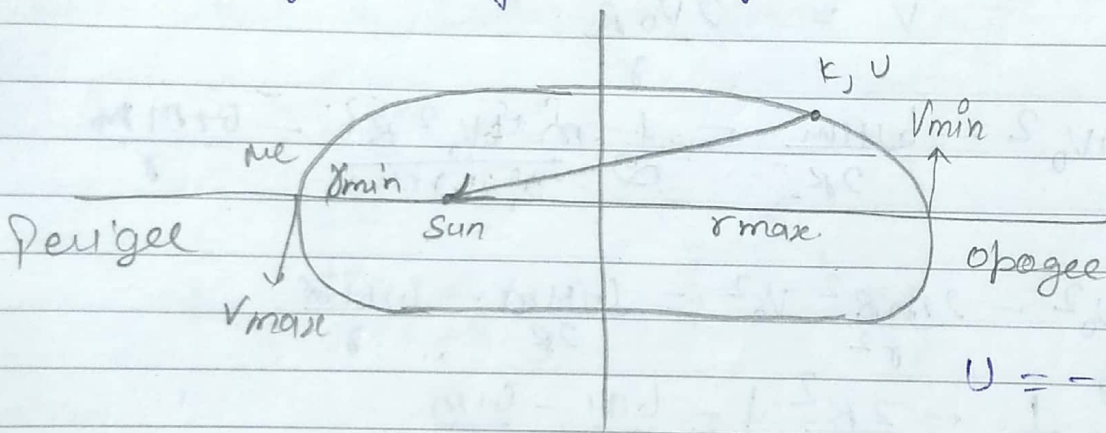
$$r = 2R$$

$$r = 3R \text{ --- max. } A$$

* Keplar's law of planetary motion:

Law:

1) Every planet revolves around Sun in elliptical orbit and Sun is always at one of the focus of ellipse



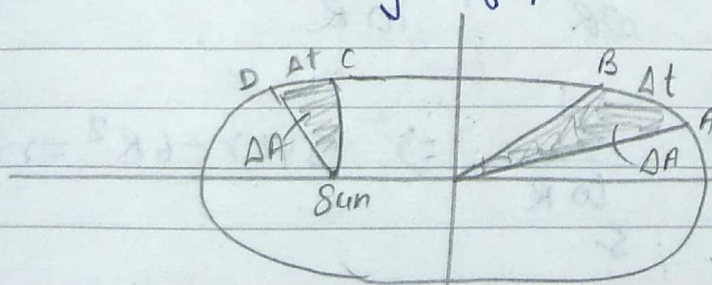
$$U = -\frac{GM_s M_e}{r}$$

τ earth about sun = 0
 $t = \text{constant}$

Law 2 line of joining of earth and Sun covers equal area in equal time interval

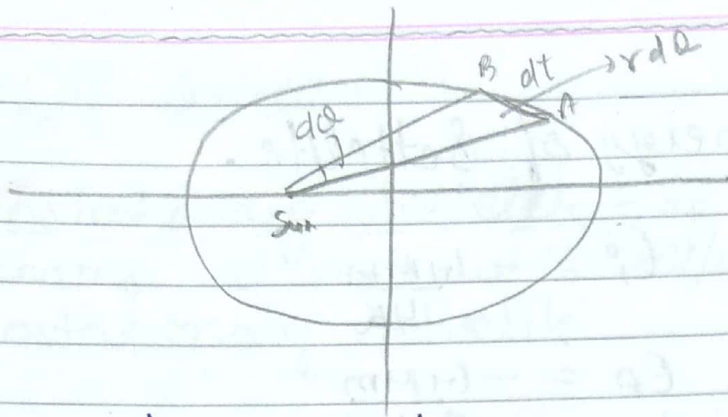
OR

Areal velocity of planet remains constant.



$$\frac{\Delta A}{\Delta t} = \frac{dA}{dt}$$

Areal velocity



$$dA = \frac{1}{2} r^2 d\theta$$

$$dA = \frac{1}{2} r^2 d\theta \Rightarrow \frac{dA}{dt} = \frac{1}{2} r^2 \frac{d\theta}{dt}$$

$$= \frac{1}{2} r^2 \omega = \frac{r v}{2}$$

$$\frac{dA}{dt} = \frac{m v r}{2 m}$$

Angular momentum.

Areal velocity

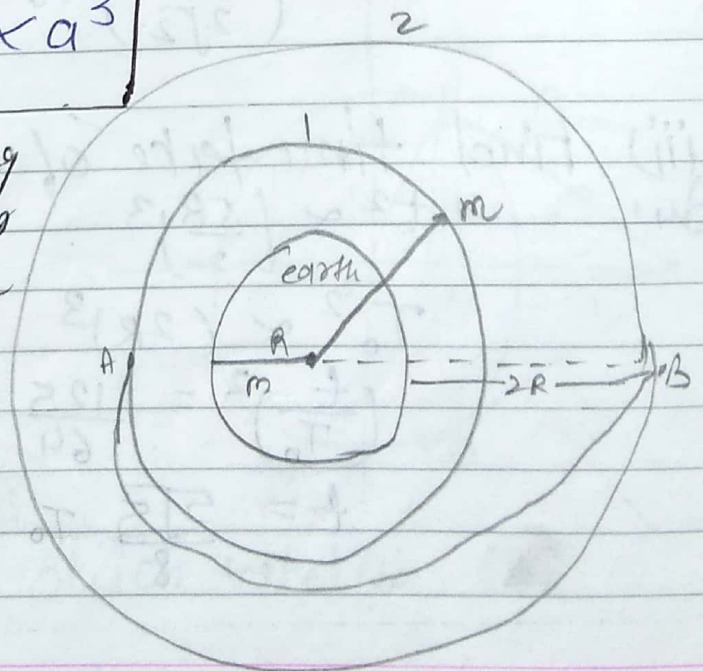
$$\frac{dA}{dt} = \frac{L}{2m}$$

mass of earth

Law-3: Time period of Planet in Elliptical orbit depends on the length of semi-major axis.

$$\text{Time Period} \rightarrow T^2 \propto a^3$$

Que: A satellite orbiting around earth in a circular path of radius $2R$ when it is at point A it is given an impulse so that it move on an elliptical path from A to B. then again start moving in a circular orbit of radius $3R$



$$P \propto T$$

$$Ex: 5-1 = 1 \text{ to } 8 \neq T$$

$$0-1 = 1 \text{ to } 10$$

Ans: find α

i) change in Energy of Satellite.

Ans:

$$E = -\frac{GMm}{2r}$$

$$E_i = -\frac{GMm}{4R}$$

$$E_f = -\frac{GMm}{6R}$$

$$E_f - E_i =$$

(ii) If time period of rotation in initial orbit was T_0 find time period in final orbit.

$$T = \frac{2\pi}{\sqrt{GM}} \cdot r^{3/2}$$

$$T^2 \propto r^3$$

$$T_0^2 \propto (2R)^3$$

$$T^2 \propto (3R)^3$$

$$\left(\frac{T}{T_0}\right)^2 = \frac{27}{8} \quad \Rightarrow \quad \frac{T}{T_0} = \frac{3\sqrt{3}}{2\sqrt{2}}$$

$$T = \left(\frac{3\sqrt{3}}{2\sqrt{2}}\right) T_0$$

(iii) Find time take of satellite from A to B.

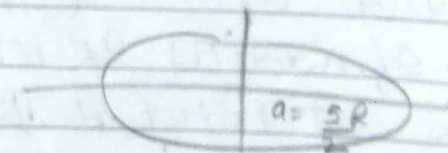
Ans:

$$t^2 \propto \left(\frac{5R}{2}\right)^3$$

$$T_0^2 \propto (2R)^3$$

$$\left(\frac{t}{T_0}\right)^2 = \frac{125}{64}$$

$$t = \frac{5\sqrt{5}}{8} T_0$$



$$t_{AB} = \frac{5\sqrt{5}}{16} T_0 \quad \text{Ans}$$

08/06/17

* Types of Settelites

i) Geostationary Settelite :- A settelite which remains at rest with respect to earth called geostationary settelite

* Its angular velocity and time period are same as that of earth

* These settelites revolve around earth in equatorial plane. or above equator.

* Their right height from earth surface approximately $h \approx 6R_e \approx 36000 \text{ km}$.

* These settelites are used for communication on particular area of the earth

* These rotates from west to east

$$h \approx 6R_e \approx 36000 \text{ km}$$

$$r = R + h$$

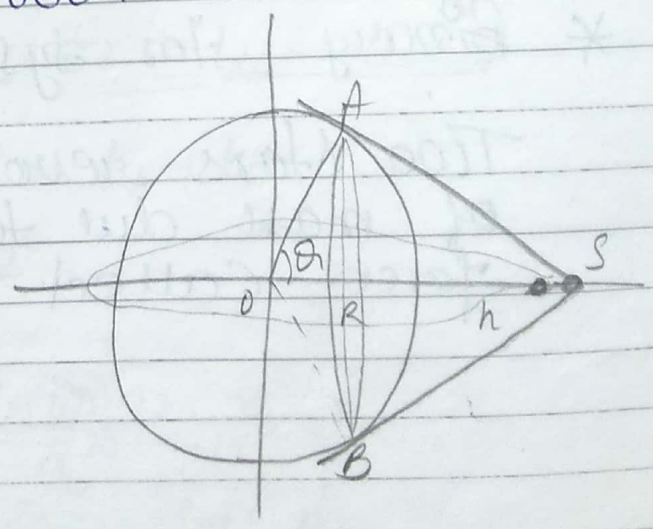
$$T = \frac{2\pi}{\sqrt{\mu}} r^{3/2}$$


$$\cos \alpha = \frac{R}{R+h}$$

$$\alpha = \cos^{-1} \left(\frac{R}{R+h} \right)$$

max latitude to which settelite can communicate.

$$\text{Co-latitude} = \cos(90 - \alpha)$$





Solid angle of Disc. $= 2\pi(1 - \cos\theta) = \Omega$

4π Solid angle have area $= 4\pi R^2$

Covered

Area

Ω

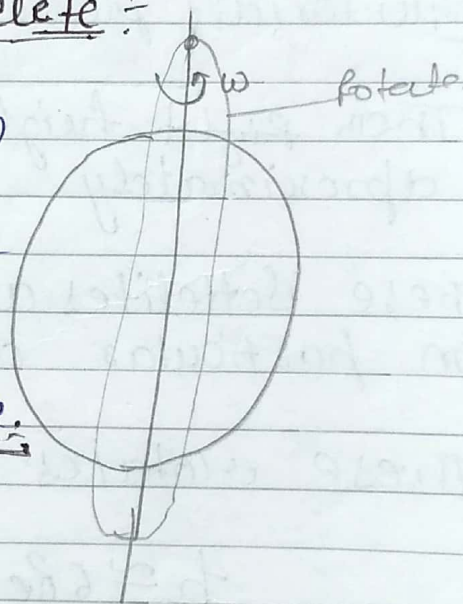
$= R^2$

$= 2\pi R^2(1 - \cos\theta)$

Area covered $= 2\pi R^2(1 - \cos\theta)$

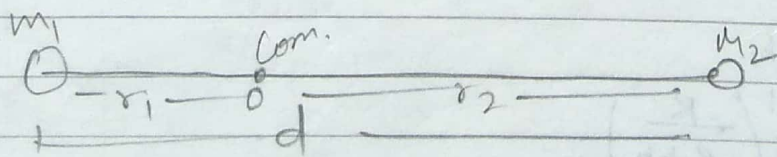
* Polar Synchronised Satellite :-

These satellites revolve in polar orbit every part of earth passes in front of the satellite twice in one rotation. These satellites are also called spy satellite.



* Binary Star System :-

Two stars revolving around their centre of mass due to their mutual gravitational force called Binary star system.



$$\frac{r_1}{r_2} = \frac{m_2}{m_1} \quad r_1 + r_2 = d$$

$$r_1 = \frac{m_2 d}{m_1 + m_2}$$

$$r_2 = \frac{m_1 d}{m_1 + m_2}$$

$$\Rightarrow f = \frac{GM_1 M_2}{d^2} \Rightarrow \frac{m_1 v_1^2}{r_1} = \frac{m_2 v_2^2}{r_2}$$

$$v_1 = \sqrt{\frac{GM_2 r_1}{d^2}}$$

$$T_1 = \frac{2\pi r_1}{v_1}$$

$$v_2 = \sqrt{\frac{GM_1 r_2}{d^2}}$$

$$T_2 = \frac{2\pi r_2}{v_2}$$

$$T_1 = T_2$$

$$\frac{r_1}{v_1} = \frac{r_2}{v_2}$$

$$\omega_1 = \omega_2$$

$$\frac{1}{\omega_1} = \frac{1}{\omega_2}$$

Kinetic Energy of system: = $K_1 + K_2$.

$$= \frac{1}{2} m_1 v_1^2 + \frac{1}{2} m_2 v_2^2$$

$$= \frac{1}{2} \frac{GM_1 M_2}{d^2} r_1 + \frac{1}{2} \frac{GM_1 M_2}{d^2} r_2$$

$$= \frac{1}{2} \frac{GM_1 M_2}{d^2} (r_1 + r_2) = \frac{GM_1 M_2}{2d}$$

$$P.E = - \frac{GM_1 M_2}{d}$$

$$T.E = - \frac{GM_1 m_2}{2d}$$

$$m_1 v_1 r_1 = m_2 v_2 r_2$$

* Angular momentum!

$$L_{\text{system}} = L_1 + L_2 = I\omega$$

$$= M_1 v_1 r_1 = M_2 v_2 r_2$$

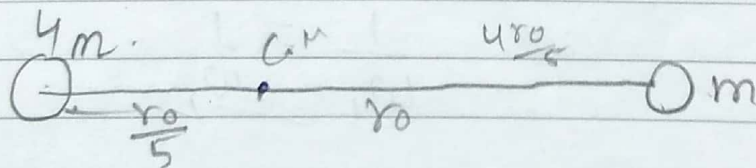
$$= I\omega = I_1\omega + I_2\omega$$

$$= (m_1 r_1^2 + m_2 r_2^2)\omega$$

$$\frac{L_{\text{system}}}{L_1} = \frac{L_1 + L_2}{L_1} = 1 + \frac{L_2}{L_1}$$

$$= 1 + \frac{M_2}{M_1} \left(\frac{r_2}{r_1} \right)^2 = 1 + \frac{M_1}{M_2}$$

Que:



Both are revolving around centre of mass in circular orbit. Find time period of rotation.

$$\frac{GMm}{r_0^2} = \frac{4m v^2}{r_0}$$

$$\frac{GM}{5r_0} = v^2$$

It is independent of position and orientation of tunnel

HW: 0-1 complete and 0-2

2π

$$T = \frac{2\pi r_0/5}{v}$$

$$T = \frac{2\pi r_0}{\sqrt{\frac{461M}{5r_0}}} = T = \frac{2\pi}{\sqrt{561M}} r_0^{3/2}$$

* Simple Pendulum:

1) $T = 2\pi \sqrt{\frac{L}{g}}$ (CCCCC Re)

$$T = 2\pi \sqrt{\frac{1}{g \left(\frac{1}{L} + \frac{1}{Re} \right)}}$$

2) $\frac{1}{L} \gg \gg \gg \frac{1}{Re}$ $T = 2\pi \sqrt{\frac{L}{g}}$

3) $L \gg \gg \gg Re$
 Simple pendulum of infinite length
 $\frac{1}{L} \ll \ll \ll \frac{1}{Re}$

$$T = 2\pi \sqrt{\frac{Re}{g}} = 2\pi \sqrt{\frac{Re^3}{61M}} \quad g = \frac{61M}{Re^2}$$

Cover
metal
crust

* Gravitational Pressure!

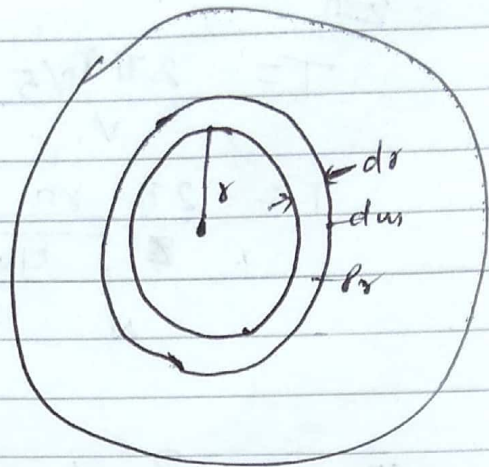
dmg

\oint

$$\int 4\pi r^2 dr \frac{G M m}{R^3} r = P_r 4\pi r^2$$

$$P_r = \frac{G M m}{R^3} \int_0^R r dr$$

$$P_r = \frac{G M m}{2R^3} (R^2 - r^2)$$



SBG STUDY